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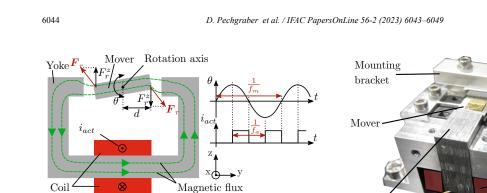


Fig. 1. Resonant rotational reluctance actuator working principle: The magnetic flux generated by the current in the actuator coil i_{act} is guided to the mover via a ferromagnetic yoke. The current is controlled in a way, that the reluctance force uphelds an oscillating motion of the mover around its rotation axis.

prototype system is introduced in Section 2, three driving strategies are derived in Section 3 to vary the injected energy into the systems mover oscillation, allowing to control the scanning amplitude. In Section 4 transfer functions for the linearized system are identified in different operating points, followed by the design of tailored controllers for each driving strategy. The corresponding closed-loop dynamics are evaluated over a wide operating range in Section 5 whereupon the paper is concluded in Section 6.

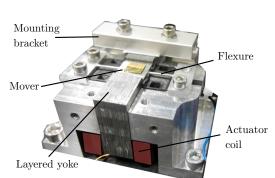
2. SYSTEM DESCRIPTION

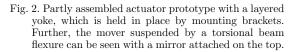
2.1 Resonant Rotational Reluctance Actuation Principle

Figure 1 shows the working principle of the resonant rotational reluctance actuator [Pechgraber et al. (2023)]. A ferromagnetic mover is suspended by a flexure structure, which enables the rotation around its rotation axis (parallel to x-axis) and restricts the other degrees of freedom (DoF). For generating a scanning motion of the mover, it is actuated by the reluctance force between the ends of the ferromagnetic yoke and the mover. In the presence of a magnetic flux (green dashed lines), the reluctance force F_r acts on the mover. The force components in z-direction F_r^z result in a net torque around the rotational axis if the mover is deflected from its equilibrium position by an angle $|\theta| > 0$.

The reluctance force is proportional to the square of the coil current i_{act} , and thus a solely attractive force [Schmidt et al. (2020)]. Hence, only pulling forces can be applied to the mover in this configuration. Such systems are usually driven by a square-wave signal with twice the oscillating frequency of the mover, which is called first-order parametric resonance [Frangi et al. (2017)]. This means that the actuator coil current is driven with the frequency f_a , which is twice the mover frequency f_m . In Fig. 1 the case is shown, in which the current is switched on exactly at the peak of the mover oscillation and switched off in the equilibrium position, avoiding a deceleration of the mover, which is called synchronized excitation [Brunner et al. (2021a); Schenk et al. (2000)].

As a consequence of the operation in first-order parametric resonance, instability occurs at the movers equilibrium





position $\theta = 0$ [Champneys (2009)]. For this reason, already small perturbations from the equilibrium position, like vibrations or an initial offset by mounting tolerances lead to the start of the mover oscillation. Therefore, the automatic startup of the system is possible by simply applying the mentioned actuation signal to the actuator.

2.2 1DoF Scanning System Prototype

The experimental evaluation of the different driving strategies of the resonant rotational reluctance actuator are carried out on a 1DoF scanning mirror prototype system [Pechgraber et al. (2023)]. The partly assembled prototype system is shown in Fig. 2. On the top, the ferromagnetic mover is visible, which is suspended via a torsional beam flexure. This torsional beam flexure enables the desired rotational movement at a mechanical resonance frequency of 223 Hz, while suppressing the other DoFs [Pechgraber et al. (2023)]. The angular position of the mover is measured by two optical proximity sensors beneath the mover. A layered ferromagnetic yoke guides the magnetic flux generated by the actuator coil to the mover. Layering of the voke reduces the unwanted phase-loss between magnetic flux and current and ohmic losses in the yoke, which are caused by eddy-currents [Csencsics et al. (2017)]. The actuator coil is driven by a custom made full-bridge pulse generator, generating the required voltage pulses for magnetizing and demagnetizing the actuator coil. A dSpace system (MicroLabBox, dSPACE GmbH, Germany) is used for the implementation of the switching logic that drives the pulse generator and for the control strategies presented in this paper.

3. DRIVING STRATEGIES

As can be seen in Fig. 1, the reluctance force generated by the actuator coil is used for upholding the oscillation of the mover. The reluctance torque acting on the mover can be approximated by [Pechgraber et al. (2023)]

$$\tau(\theta, i_{act}) = -c_1 \theta(t) i_{act}^2(t) \tag{1}$$

with the current in the actuator coil $i_{act}(t)$, the mover deflection angle from the equilibrium position $\theta(t)$ and a positive fitted constant c_1 . By integrating the torque over

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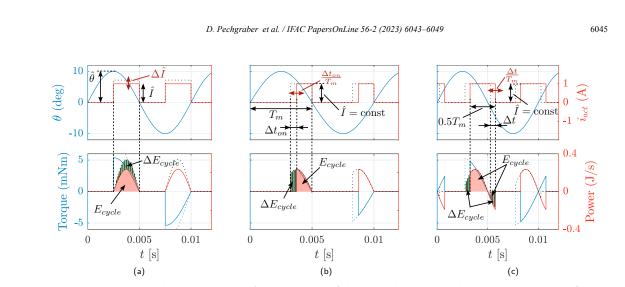


Fig. 3. Three actuator coil driving strategies for the variation of the injected energy into the system. The mover deflection angle θ with idealized actuator coil current i_{act} and the respective mover torque is shown in the case of \hat{I} -control (a), t_{on} -control (b) and phase-control (c). The area under the power curves can be interpreted as the injected energy per actuation cycle (E_{cycle}) .

the traveled angle θ the energy injected into the system in one actuation cycle can then be calculated by [Brunner et al. (2021b)]

$$E_{cycle} = \int_{\substack{\theta(0)\\\theta(0)\\\theta(T_m)\\\theta(0)\\\theta(0)}}^{\theta(T_m)} \tau(\theta, i_{act}) d\theta$$

$$= \int_{\substack{\theta(0)\\\theta(0)}}^{\theta(T_m)} - \left(c_1\theta(t)i_{act}^2(t)\right) d\theta.$$
(2)

In steady state operation for a constant ocillation amplitude the injected energy has to compensate the energy that is dissipated via damping in this period of time. In order to further control the oscillation amplitude of the system, E_{cycle} has to be varied by a controller. Figure 3 shows the three driving strategies, enabling the manipulation of the injected energy. These are the variation of either the input current amplitude, the actuation pulse duration, or the phase between the input and the mover angle. The plots show the mover deflection angle θ with idealized actuator coil currents i_{act} and the respective torque on the mover according to (1). The integration variable in (2) is the mover angle θ , and not the time t. To better understand the way the energy is injected into the system, the injected power over time is additionally plotted.

The first driving strategy in Fig. 3a shows the variation of the current pulse amplitude \hat{I} , referred to as \hat{I} -control. In this case, the actuator coil is switched on exactly at the peak of the mover oscillation $\hat{\theta}$ and turned off at the zero-crossing. By assuming ideal rectangular shaped actuator current pulses, (2) can be evaluated to

$$E_{cycle}^{\hat{I}} \approx \int_{\hat{\theta}}^{0} - \left(c_1 \theta \hat{I}^2\right) d\theta = \frac{1}{2} c_1 \hat{\theta}^2 \hat{I}^2 , \qquad (3)$$

in the case of \hat{I} -control. This energy can be interpreted as the area below the power curve in Fig. 3a (red area). If the current amplitude is changed by $\Delta \hat{I}$, a change of injected energy per actuation cycle ΔE_{cycle} is the consequence (green shaded area).

The second driving strategy is the variation of the current pulse duration, i.e. duty-cycle, while leaving the current amplitude \hat{I} constant. In Fig. 3b the off time is fixed at the zero-crossing and the on time is varied. For the control implementation in the next section the on time is normalized to the actuation period time T_m , which is a half mover oscillation period. Therefore, the respective control variable is $\frac{t_{em}}{T_m}$. The injected energy can be expressed with the control variable, by evaluating (2) according to Fig. 3b, yielding

$$E_{cycle}^{t_{on}} \approx \int_{\theta(t_{on})}^{0} - \left(c_1 \theta \hat{I}^2\right) d\theta = \frac{1}{2} c_1 \hat{I}^2 \theta^2(t_{on}) \,. \tag{4}$$

If a sinusoidal motion $\theta(t) = \hat{\theta} \sin(\omega_s t)$ with the angular frequency $\omega_s = \frac{\pi}{T_m}$ is assumed, the injected energy can be expressed by

$$E_{cycle}^{t_{on}} \approx \frac{1}{2} c_1 \hat{I}^2 \hat{\theta}^2 \sin^2 \left(\pi \frac{t_{on}}{T_m} \right), \tag{5}$$

for $\frac{t_{cn}}{T_m} \in [0, 1]$. As in the \hat{I} -control case the area under the power curve in Fig. 3b (red area) is the energy per actuation cycle E_{cycle} , and the energy difference ΔE_{cycle} (green shaded area) results from the change of the on time $\frac{\Delta t_{cm}}{T_m}$. For the control implementation the sign of the control variable has to be inverted, because in the range $\frac{t_{cm}}{T_m} \in [0.5, 1]$ the injected energy function (5) has a negative slope. The system could also be driven in the range $\frac{t_{cn}}{T_m} \in [0, 0.5]$, but that causes losses by deceleration of the mover in the region before the amplitude.

Figure 3c shows the third case, in which the pulse duration and the current amplitude are constant, but the relative phase between the current pulse and the mover oscillation is varied. The introduced time shift Δt is again normalized

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to the actuation period T_m for the control implementation, hence the control variable is $\frac{\Delta t}{T_m}$. With a sinusoidal motion, trigonometric identities and by neglecting variations of T_m and $\hat{\theta}$, the evaluation of (2) leads to

$$E_{cycle}^{\Delta t} \approx \int_{\theta\left(\frac{T_m + \Delta t}{2}\right)}^{\theta\left(T_m + \Delta t\right)} - \left(c_1\theta\hat{I}^2\right)d\theta$$
$$= \frac{1}{2}c_1\hat{I}^2\hat{\theta}^2 \left[\sin^2\left(\frac{\pi}{2} + \frac{\Delta t}{T_m}\pi\right) - \sin^2\left(\pi + \frac{\Delta t}{T_m}\pi\right)\right] \quad ^{(6)}$$
$$= \frac{1}{2}c_1\hat{I}^2\hat{\theta}^2 \cos\left(2\pi\frac{\Delta t}{T_m}\right),$$

for $\frac{\Delta t}{T_m} \in [0, 1]$. In comparison to the other two driving strategies, the injected energy in an actuation cycle can get negative in the admissible range, so energy can also be withdrawn from the system. For the implementation the sign of the control variable has to be inverted again, because in the targeted driving range $\frac{\Delta t}{T_m} \in [0, 0.5]$ the injected energy function has a negative slope.

4. SYSTEM IDENTIFICATION AND CONTROL DESIGN

Because the system is nonlinear as can be seen in the derived functions for the injected energy in the previous section, linear control design procedures can only be applied for the linearized system around an operating point [Brunner et al. (2021b)]. For this reason, this section deals with the system identification for each driving strategy in different operating points and the control design.

4.1 System Identification

To enable the identification at specific oscillation amplitude setpoints $\hat{\theta}^d$, the identification is done in a closed-loop configuration. In Fig. 4 a block-diagram of the entire control system is shown. It consists of the actuator itself, the custom-made pulse generator for driving the actuator coil, sensor electronics, and the software parts implemented in the rapid prototyping system. The switching logic is responsible for the correct timing and amplitude of the actuator coil pulses, in dependence of the zero-crossings of the sensed mover angle θ and according to the control input of the respective driving strategy. The switching logic together with the zero-crossing detection acts as a phase locked loop (PLL).

For the identification a PI-controller is used, whose parameters are empirically tuned to obtain a stable oscillation amplitude at the desired setpoint $\hat{\theta}^d$. To detect the amplitude of the mover oscillation $\hat{\theta}$, the sensed mover angle is filtered with a 6th order Chebyshev low-pass filter with a cut-off frequency of 600 Hz. This reduces the remaining noise on the measured signal and ensures a correct detection of the amplitude values. The filter also adds a dead-time of $\approx 775\mu s$ at the scanning frequency of 223 Hz. However, a new amplitude value is gathered only once every mover oscillation period of 4.5ms, so the added dead-time is acceptable. For the system identification a network analyzer (Analog Discovery 2, Digilent, USA) is connected to the system according to Fig. 4. The sampling time of the software system is $T_s = 25 \,\mu s$, therefore the delay introduced by the analog-digital converters (ADCs) and digital-analog converters (DACs) of the system can be neglected and the plant transfer function G(s) can be approximated with the measured signals by

$$G(s) = \frac{\hat{\theta}(s)}{u(s)} \approx \frac{\gamma(s)}{\xi(s)}, \qquad (7)$$

with $u(s) \in \left\{ \hat{I}(s), \frac{t_{en}}{T_m}(s), \frac{\Delta t}{T_m}(s) \right\}$. $\gamma(s)$ and $\xi(s)$ represent analog signals after the DACs of $\hat{\theta}$ and u respectively.

In Fig. 5 the result of the system identification is shown. For each driving strategy (blue, green, black) three amplitude set points (2 deg, 5 deg, 8 deg) are investigated, marked with dotted, dashed and solid lines in the plot. In t_{on} -control and phase-control the injected energy functions (5) and (6) have a negative slope in the area of operation and thus the phase is inverted compared to \hat{I} -control. However, as the sign of the control variable is inverted in the implementation, the phase in these cases is shifted by -180 deg in Fig. 5.

The system shows 1st order low-pass behaviour with additional dead-time. For the set point $\hat{\theta}^d = 2 \deg$ the phase is $-90 \deg$ also for very low frequencies of below 1 Hz, what would suggest a pure integrator. However, if the integral gain of the implemented controller is turned off, a steady-state control error occurs, ruling out pure integral behaviour of the system. The phase-drop in all cases is mainly dominated by the dead-time introduced by the fact, that a new amplitude value is detected only once every oscillation cycle in the amplitude detection block of Fig. 4. The fitted dead-time in the phase-plot of Fig. 5 (red dashed line) corresponds to $T_{dt} = \frac{1}{223\text{Hz}} = 4.5 \text{ ms.}$ The largest magnitude difference between the lowest and highest operating point is visible in \hat{I} -control.

4.2 Control Design

To design a robust controller for each driving strategy, the identified plant transfer functions G(s) with the largest setpoint are used. Otherwise the open-loop cross-over frequency would be shifted to the right for higher operating points, reducing the respective phase-margin or even leading to an unstable system. By using the fitted dead-time from Fig. 5 and linear fits for the -20 dB slopes in Fig. 5 for each of the three driving strategies a PI-controller with the structure

$$G_{PI}(s) = k \frac{1+sT}{s} \tag{8}$$

is designed. The phase-lag of the identified dynamics are given by

$$\arg\left(G(j\omega)\right) = -\frac{\pi}{2} + \arg\left(e^{-sT_{dt}}\right) = -\frac{\pi}{2} - \omega T_{dt} \,. \tag{9}$$

If zero phase-lag from the controller is assumed, the maximum open-loop cross-over frequency for a desired phase-margin of $\Phi = 60 \deg$ can then be calculated by

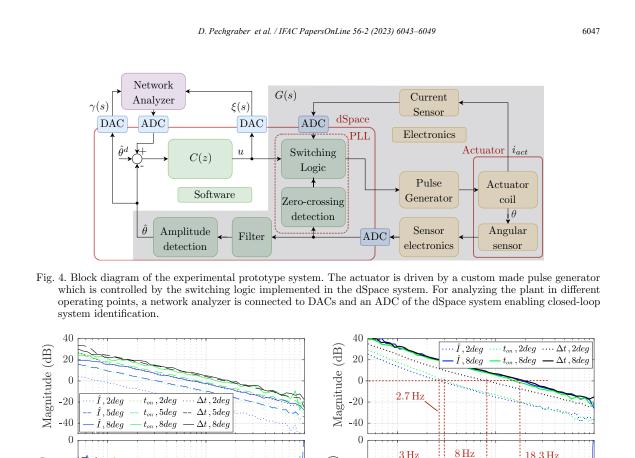
$$\omega_c = \frac{\frac{\pi}{2} - \Phi}{T_{dt}} = 115.18 \,\text{rad/s} \equiv 18.3 \,\text{Hz}\,.$$
(10)

The zero of the PI-controller (8) is placed a factor 10 below this frequency to justify the assumption of zero phaselag at the cross-over frequency, so $T = \frac{10}{\omega_c}$. Afterwards

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Phase (deg)

-90

-180

-270

 10^{0}

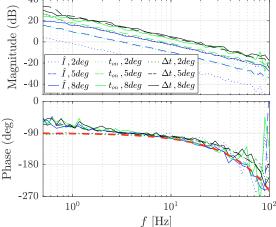


Fig. 5. Results of the system identification showing the plant transfer function for each driving strategy in 3 different operating points $\hat{\theta}^d$. The fitted time-delay $T_{dt} = 4.5 \,\mathrm{ms}$ is illustrated by the red dashed line in the phase plot.

Table 1. Coefficients for the implemented PIcontroller

Strategy	$_{k}$	T	$\frac{\omega_c}{2\pi}$	Unit
\hat{I} -control	29.829	0.0868	18.3 Hz	A/deg
t_{on} -control	15.0499	0.0868	18.3 Hz	$1/\deg$
phase-control	11.3962	0.0868	$18.3 \mathrm{Hz}$	$1/\deg$

k is adjusted, so that the loop gain crosses the 0 dB line at the desired frequency. The given phase-margin results in the controller parameters from Table 1. For clarification also the units of the controller gains are given. The controller outputs are subject to certain physical limits, like $\hat{I} = 0$ A as lower bound for \hat{I} -control and $\frac{t_{on}}{T_m} \in [0.5, 1]$ or $\frac{\Delta t}{T_m} \in [0, 0.5]$ as bounds for t_{on} -control and

Fig. 6. Open-loop transfer functions with implemented PIcontroller for each driving strategy. The controller are designed for the highest operating point ($\hat{\theta} = 8 \deg$), therefore the open-loop crossover frequency changes significantly for the lower operating point ($\hat{\theta} = 2 \deg$).

 10^{1}

f [Hz]

 $60 \deg$

 10^{2}

phase-control. Therefore the outputs are limited to their bounds by software and an algorithm is implemented to avoid integral wind-up.

For verification of the control design, the open-loop transfer functions with the designed PI-controllers are measured and are shown in Fig. 6. In the operating point $\hat{\theta}^d = 8 \deg$ the open-loop transfer function of all three driving strategies matches well with the expected crossover frequency. Also the phase-margin is well in the targeted area of Φ = 60 deg. However, a large shift to lower cross-over frequencies is visible for the operating points with smaller amplitude. Especially for \hat{I} -control and t_{on} control the cross-over frequency is reduced by a factor of 6.8 and 6 respectively, whereas for phase-control it is

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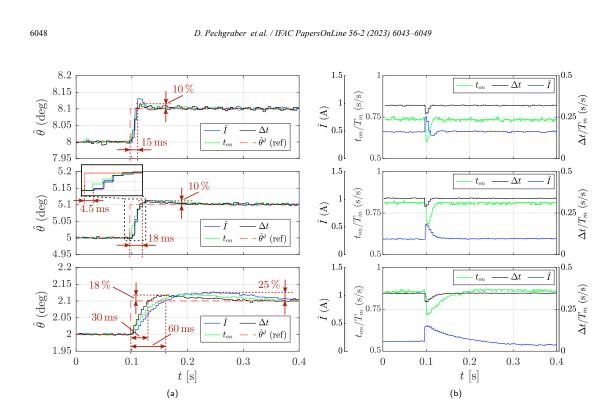


Fig. 7. Dynamic evaluation of the control system for each driving strategy by a step-response on the reference input. On the left the oscillation amplitude is shown together with the step on the reference input (a), and on the right side the corresponding control action is plotted (b).

only a factor of 2.3. Closed-loop measurements for the operating point $\hat{\theta}^d=8$ deg reveal a closed-loop bandwidth of 29 Hz for \hat{I} -control and phase-control and 22 Hz for t_{on} -control (data not shown). In the lowest operating point $\hat{\theta}^d=2$ deg phase-control has a closed-loop bandwidth of 11 Hz, whereas the other two driving strategies only have ≈ 3.5 Hz.

5. EXPERIMENTAL RESULTS

To evaluate how the different driving strategies affect the closed-loop dynamics of the system in the time-domain, a comparison of responses to reference input steps is performed. As in the open-loop transfer functions from Fig. 6, the same controller which is designed for the operating point $\hat{\theta}^d = 8 \deg$ is applied for every operating point, to enable a fair comparison between the driving strategies.

Figure 7 shows a step-response with 0.1 deg amplitude in each of the identified operating points. On the left side the oscillation amplitude is plotted together with the step on the reference input and on the right side the corresponding control action is shown for each driving strategy. For $\hat{\theta}^d = 8 \text{ deg}$ the dynamic performance is very similar for all driving strategies, because the controllers are tailored for this operating point. They show a rise-time of approximately 15 ms, with an overshoot of $\approx 10 \%$. The performance does not change much for $\hat{\theta}^d = 5 \text{ deg}$ with a rise-time of 18 ms and similar overshoot. In the zoomed area of the middle plot, the delay of ≈ 4.5 ms brought into the system by the amplitude detection can be seen.

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As expected from the measured open-loop transfer functions in Fig. 6, the biggest difference between the driving strategies is visible in the lowest operating point $\hat{\theta}^d = 2$ deg. In this case, the difference between phasecontrol and the other strategies is clearly visible. \hat{I} -control delivers with 60 ms a factor 2 higher rise-time than phasecontrol, and a 7% higher overshoot. t_{on} -control shows a rise-time of 50 ms, which is a factor 1.7 higher as compared to phase-control, with comparable overshoot. The reason is the larger magnitude variation of the plant transfer function and the resulting lower open-loop cross-over frequency for the lower operating points for \hat{I} -control and t_{on} -control.

In summary, three different driving strategies for the oscillation amplitude control of a resonant rotational reluctance actuated scanning mirror system are developed and successfully demonstrated on an experimental prototype system. By varying either the current input amplitude, the pulse duration or the phase between actuation pulse and mover angle the amplitude of the scanning mirror system can be successfully controlled, whereby phase-control shows the smallest degradation of closed-loop dynamics for smaller operating points and outperforms the other driving strategies by at least a factor of 1.7 in rise-time during a step-response.

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6. CONCLUSION

In this paper three different driving strategies for the oscillation amplitude control of a resonant rotational reluctance actuated scanning mirror system are designed and implemented on an experimental prototype system. Formulas for the injected energy into the system per actuation cycle are derived for each driving strategy based on idealized actuator signals. Due to the non-linear nature of the controlled system, PI-controllers are designed for the linearized plant around an operating point for each of the driving strategies and compared regarding their closedloop dynamics. For three amplitude choices covering a wide operating range, phase-control shows a rise-time of 30 ms to a step on the reference input at worst. This outperforms \hat{I} -control by a factor of 2 and t_{on} -control by a factor of 1.7. Phase-control with a linear PI-controller is therefore the most promising control strategy for the oscillation amplitude control of the resonant rotational reluctance actuated scanning mirror system in terms of a wide range of uniform operation.

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